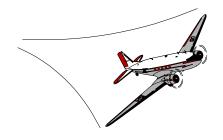
SPECIAL AIRWORTHINESS INFORMATION BULLETIN

Aircraft Certification Service Washington, DC





U.S. Department of Transportation

Federal Aviation Administration

CE-03-16 January 6, 2003

We post SAIBs on the internet at "av-info.faa.gov"

This is information only. Recommendations are not mandatory.

Introduction

This Special Airworthiness Information Bulletin (SAIB) provides safety information **to Schempp-Hirth Model Nimbus-3/24.5**, all serial numbers, sailplane owners on the following:

- Extension of Service Life to 12,000 hours, and
- Supplementing the Maintenance Manual with a special multi-stage inspection program.

Background

Schempp-Hirth has developed Technical Note No. 286-34 that extends the service life and supplements the Maintenance Manual. We have attached the applicable service information.

Safety Issue

The German Airworthiness Authority, Luftfahrt-Bundesamt, has issued Airworthiness Directive (AD) 2002-357, requiring compliance of this Technical Note. The FAA does not issue ADs for extension of service life.

General Information

We are providing this information as a courtesy to operators of the Schempp-Hirth Model Nimbus-3/24. We do not issue ADs for sailplanes that do not have a U.S. type certificate. The FAA will not track or enforce this issue. Furthermore, the FAA will not use an SAIB to track all safety related items for non-type certificated products.

For Technical Information Concerning These Safety Issues Contact

Schempp-Hirth Flugzeugbau GmbH, Postfach 14 43, D-73222 Kirchheim/Tech, Federal Republic of Germany; Telephone 49.7021.7298-0; Facsimile 49.7021.7298-199.

Further Information Contact

Tom Knauff, United States Schempp-Hirth Dealer, 3523 South Eagle Valley Road, Jullian, PA 16844; Telephone (814) 355-2483; Facsimile (814) 355-2633

Greg Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri, 64106; telephone (816) 329-4144; Facsimile (816) 329-4090, email; mike.kiesov@faa.gov.

SCHEMPP-HIRTH		Technical Note	Blatt: 01
Flugzeugbau GmbH. Kirchheim/Teck		No. 286-34	Blattz.: 02
SUBJECT: AFFECTED:	b) Suppleme	of the service time ents of the Maintenance Manual odel "Nimbus-3/24.5 (ATC-No. 286 umbers –	3)
URGENCY:	On reaching but not later	a service time of 6000 flight hour than on the occasion of the next a	s, annual inspection.
REASON:	spar sect of GFRP provided evidence then to be	Its of fatigue tests (subsequently clions) have demonstrated that the / CFRP sailplanes may be extend the airworthiness of each individud by a special multi-stage inspectie incorporated into the Maintenan	time in service ed to 12000 hours, al aircraft is on program, which is ce Manual.
ACTIONS:	Maintena Revisions o	on of maintenance by supplementi ince Manual. If the Maintenance Manual ges dated September 2002)	ng the
	Page 2/3 1 17A 22 32 32A	Title - Record of revisions - Table of contents - Play in the tailplane attachmer - Hinge moments and weights of - Special inspections of the airfr - Special inspections of the airfr	of control surfaces ame
	an inspect	one of the various limits in service in service to be conducted according the service obtained from the manufac	ng to an rvice time"

SCHEMPP-HIRTH	Technic	cal Note	Blatt:	02
Flugzeugbau GmbH. Kirchheim/Teck	No. 2	86-34	Blattz.:	02
MATERIAL:	Not affected			
WEIGHT:	Not affected			and the state of t
C/G POSITION:	Not affected			
				,
Note:	Accomplishment of log book by a licens	action to be checked a sed inspector.	and entered i	n the
Kirchheim/Teck, Septe	mber 23, 2002	LBA-approved: The german original of the been approved by the last of the last	5A under the da 5 SEP. 2002	ate
Issued: (H. Treiber	ils	and is signed by Mr The translation into Englibest knowledge and judge	lish has been d	

Blatt-Nr. 08

Übersicht über Schempp-Hirth Technische Mitteilungen und LBA-Lufttüchtigkeitsanweisungen

Durchgeführt page complied REGISTRATION NO.: than on the occasion of the next Beim Erreichen von 6000 Flug-On roaching a service time of 6000 flight hours, but not later stunden, spätestens bei der nächsten Jahresnachprüfung Kennzeichen: annual inspection. Urgency Termin Werk-Nr.: SUMMARY OF SCHEMPP-HIRTH TECHNICAL NOTES AND LBA-AIRWORTHINESS DIRECTIVES S/N: Erhöhung der Betriebszeit auf 12.000 Stunden Extension of service time up to 12.000 hours MODEL/VARIANT: Muster/Baureihe: Gegenstand Subject - alle Werknummern -Baureihen und Werk-Nr. 286 all serial numbers -DATA SHEET betriff Nimbus-3/24.5 Kennblatt-Nr. Model affected: S TM-Nr. Ausgabe Datum TN-No. date of issue 286-34 23.09.2002 Nimbus-2 Ausgabe Datum 14.11.2002 Date of issue BA-LTA-Nr. LBA-AD-No. 2002-357 Muster: MODEL

The results of the inspections are to be recorded in an inspection report, wherein comments are required for each inspection step.

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If the inspections are carried out by a certified repair station, a copy of the records must be sent to the manufacturer for evaluation.

The mandatory annual inspection is not affected by this regulation

Special inspections of the airframe

Inspection program for the extension of the service time

General

The results of fatigue tests subsequently carried out on wing spar sections have demonstrated that the service time of GFRP/CFRP saliplanes and powered saliplanes may be extended to 12,000 hours, if for each individual aircraft (in addition to the obligatory annual inspections) the airworthiness is demonstrated according to a special multi-step inspection program, particularly with regard to the service life.

Dates

When the saliplane (or the powered saliplane) has reached a service time of 6000 hours, an inspection must be done in accordance with the inspection program mentioned under item 3 further below.

If the results of this inspection are satisfactory or if any defects found have been duly repaired, the service time of the sailplane (or powered sailplane) is extended by another 3000 hours to a total of 9000 hours (first step).

The afore-said inspection program must be repeated when the sailplans (or the powered sailplane) has reached a service time of 9000 hours.

If the results of this inspection are satisfactory or if any defects found have been duly repaired, the time in service may be extended by another 1000 hours to 10.000 hours (second step) after a further 1000 hours inspection to 11.000 hours (third step), and finally – after another 1000 hour inspection – to 12.000 hours (fourth step).

- The relevant inspection program may be obtained from Schompp-Hidh Flugzeugbau GmbH.
- The inspections may only be accomplished by the manufacturer or by a certified repair station.

- 17A -

If the values of the table on page 21 are exceeded, it will be necessary to add an additional balance weight forward of the hinge line as follows:

- After repair work in the area of the repair.
- After refinishing in the refinished area distributed along its whole length (if possible)

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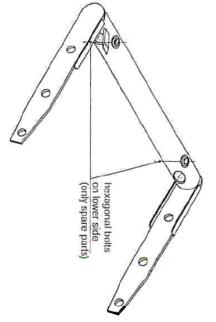
After complete repainting – distributed along the whole length of the component (if there was no mass balance attached) or near or next existing mass balance.

Exceeding the value in the above weight table because of additional mass balance is permissible up to 15%, provided the residual moment within the tolerance.

Play in the tailplane attachment bracket

If an excessive vertical play is found at the tips of the horizontal stabilizer, then the adjustable bushings of the elevator actuating bracket should be tightened such (using a 4 mm Allen key) that the tailplane will just slide onto its locating pins.

ELEVATOR ACTUATING BRACKET



Technical Note No. 286-34

Nimbus 3/24.5

MAINTENANCE MANUAL

- 2/3 -

	AMENDMENT LIST (log of revisions)		
No.	No. Reference/short title	Page	Date
12.	cal Note No. 286-25 al tilt-up instrument panel	12, 20A	April 1989
13.	Technical Note No. 286:26 Tow releases "F 85" and "G 88"	12, 20	February

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	Technical Note No. 286-34 a) Inspection program for extending the service time up to 12.000 hours b) Supplements of the Maintenance Manual	Modification Bulletin No. 286-27 Modifications on S/N 37 only (lateral control, fin/rudder unit)	Vechnical Note No. 286-27 Wing tip extensions for 25,5 m span (optional)	Teelmical Note <u>No. 286-26</u> Tow releases "F: 85" and "G 88"	Technical Note No. 286-25 Optional tit-up instrument panel
	1, 17A, 22, 32, 32A	16A, 21, 35	10, 16A, 21, 25, 26	12, 20	12, 20A
	September 2002	January 1998	March 1990	February 1990	April 1989

Nimbus-3/24.5
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AINTENANCE MANUAL

-1

	4 10 00	2.9 2.10 3.	2.1 2.2 2.3 2.4 2.5 2.6 2.7	1,	Table
Electrical wiring diagram Control circuits Appendix Polar diagram Optimum airspeeds and flap settings Repair procedures for a CRP/FRP-sailplane	Equipment list Cockpit placards and symbols Special inspections of the airframe	Control surface moments and weights Caring for a CRP/FRP sailplane Center of gravity determination	Maintenance Prescribed Checks Regular Maintenance Adjustment Data Play in the control circuits Play in the wing attachment fittings Play in the tailplane attachment fitting Damage Replacement the wing locating bearings Removal & re-installation of tow hook	Storage, Transport & Rigging Storage, Hangaring & Towing Rigging De-rigging	Table of contents Amendment List
34 33 5	30 32 32 32	25 J	11 13 16 17 17 17 17 18	3 5	Page 2

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